MODELS: Beechcraft C17R (Army UC-43E), SC17R

T.C. NUMBER: ATC 604

## I - SPECIFICATIONS PERTINENT TO ALL MODELS LISTED BELOW:

98 gallons (3 tanks: fuselage front 28 gallons (+32), fuselage rear 47 gallons (+55) and lower wing 23 gallons (+24) 6 gallons (-13) 4 maximum (front +19, rear +60.5) Fuel Oil

No. passengers (See NOTE 3)

See NOTE 3)

Baggage 125 lbs. maximum (+86) (See NOTE 3)

Spec. basis Approved Type Certificate No. 604

C17R-73 and up manufactured prior to 9/30/39

and all AAF Nos. of UC-43E eligible. Approval expired as of 9/30/39.

EQUIPMENT: (Datum is lower wing leading edge) (\* Means net increase)

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Class I:
      A. Landplane
101. Engine ring cowl
                                                                                  28 lbs.
               Starter (Eclipse E-80)
Generator - 15 amp (Eclipse G-1)
Battery (Exide 6TS-13-1) (Under baggage
      102.
                                                                                  20 lbs.
      103.
                                                                                   15 lbs.
      104_
               compartment)
                                                                                  36 1hs.
      105. Control column - Swing-over or "T"
               type (Inspector should report type
               installed)
               Oil radiator (Harrison)
Propeller - fixed pitch metal
6.50-10 or 7.50-10 wheels (Hayes)
      106.
                                                                                    8 lbs.
      107.
                (Wheels must be approved for at least
               1950 lbs. per wheel and must be placarded for 7.50-10 6-ply tires)
                                                                                  75 lbs.
               10x3 tail wheel
      B. Seaplane - Items 101 thru 106, 203 and 210 PLUS
      151. Edo 39-4000 float installation with
               water rudder
                                                                                527 lbs.
Class II:
      200. Miscellaneous items as noted in approved
               weight and balance report.
      201. Miscellaneous extra instruments
              Propeller - Hamilton Standard adj. metal
Propeller - Hamilton Standard hub
      202.
      203.
               2D30 with:
              Blades 6095A-12 to 6095A-17, inclusive, or blades 6101A-18 to 6101A-23,
              inclusive, or blades 6167A-12 to 6167A-17, inclusive. Diameter 102 in. maximum, 96 in. minimum.

Special landing gear complying with Canadian requirements for ski
              installation
     205.
              Landing lights
               (a) Under lower wing (Grimes)
(b) In wheel well (Hall)
(c) S & M
                                                                               3.5 lbs. each (+18)
3.5 lbs. each (+37)
                                                                                Use actual weight
     206.
              Plares (Brawing I-17004)
Radio, variable (location and actual
weight of installation includes
              controls, etc., or name of manu-
facturer if already installed and
included in certificated weight empty
              should be noted on inspector's report.)
(Drawing I-17002 shows approved
              locations.)
     208.
              Shielding
              (a) Breeze
(b) Packard
                                                                                 15 lbs.
                                                                                 10 lbs.
     209.
              Extra door on right side
                                                                                  6 1bs. *
              Extra 23 or 25 gallon fuel tanks in
wings, upper (+45.5) and lower (+24)
(Any combination of fuel tanks include
     210-
                                                                                 20 lbs. each
              removal of the fuselage tanks may be
              used subject to inspector's approval
              and provided weight and balance accounts for actual installation.)
     211.
              Oil tanks
             (a) 7 gallons
(b) 9 gallons
(c) 10.3 gallons
Generator - 25 amp (Eclipse)
                                                                                                       (-13)
                                                                                                       (<del>-</del>13)
                                                                                                       (-13)
     212.
                                                                                 20 lhs.
              Golf bag compartment (Drawing I-17006).
              In bay aft of, and opens into, baggage compartment. Maximum capacity 70 lbs. golf bag and clubs only.
              Parachutes or parachute type chairs.
    214.
    215.
             Seaplane corrosion protection
                                                                                20 lbs.
             Pressure fire extinguisher (Drawing I-17001) (Container at +14)
Metal tank cover (Drawing I-17003)
Camera well (Drawing I-17006)
    216.
    217.
    218.
             Ballast installation
Special tool compartment in tail,
    219.
                                                                               Maximum 25 lbs. (+210)
    220.
              including tools
                                                                                25 lbs.
                                                                                                      (+218)
    221. Camera installation (Drawings B17E015,
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B17E016, and B17E902):
                  Camera and viewfinder 110 lbs. (+51);
                  operator's stool 4 lbs. (+65);
                  25 gallon fuel tank in baggage compartment 15 lbs. (+86). Remove: rear seat 31 lbs. (+62); cover plates for camera and viewfinder 3 lbs.
                   (+51) and rear fuselage fuel tank (+55)
          222. Skis
                  (a) Pollack P-4600
          B. Seaplane: Items 205 thru 221
    Class III:
          301. 24 in. streamline wheels (Hayes)
                                                                             75 lbs.
                  with 6-ply tires
                   (Wheels must be placarded for 6-ply
                  tires)
                  Puncture proof tubes
                                                                               9 lbs.*
                  (a) 7.50-10
(b) 24 in. streamline
                                                                               5 lbs.*
                                                                            No change in weight
          303.
                  Tail wheel steering device
          304.
                  Electric motor and installation for
                  auxiliary to manually operated
                                                                               8 1bs. *
                  retracting gear (25 amp fuse required)
     NOTE 1. The following placards were required on serial No. 74 when
          originally certificated:
(a) At the fuel valve: "USE MAIN FUSELAGE TANKS FOR TAKE-OFF AND
                 LANDING. USE FUEL OUT OF AUXILIARY LOWER WING TANKS BEFORE
                  EMPTYING MAIN TANKS."
           (b) At rear fuselage fuel tank: "AS 4-PLACE 47 GALS., AS 5-PLACE
                   27 GALS."
           (c) Baggage placarded to 75 lbs.
          Similar instructions must be placed in subsequent airplanes when the
          actual weight and balance report is based on any but the most adverse (from balance standpoint) use of the fuel. Inspector's report and
           weight and balance report should specify restrictions applicable.
     NOTE 2. Eligible for export as follows, subject to inspection for equipment specified in Chapter XII of Inspection Handbook: (9/30/39)
           (a) Canada
                  - Landplane
                 - Seaplane - maximum standard weight 4000 lbs.
- Skiplane - not eligible. However, structure complies with Canadian requirements for ski installation when approved
                    special landing gear (Item 204) is installed.
                All other countries except Australia
     NOTE 3. Five place type must be placarded for 75 lbs. when 3 passengers occupy rear seat unless more is substantiated in weight and balance
           report. Baggage includes anchor and rope 25 lbs. when carried.
II - MODEL C17R (UC-43E), 5 PCLB:
                                          Wright R-975E-3 or R-975E-2
      Engine
     Placard limits
          with fixed metal or adj. metal propeller

Maximum, except take-off 33.0 in.Hg., 1800 rpm (330 hp)

or 31.5 in.Hg., 1900 rpm (330 hp)

or 29.5 in.Hg., 2000 rpm (330 hp)

or 28.5 in.Hg., 2100 rpm (330 hp)

Take-off (one minute)

or 35.0 in.Hg., 2200 rpm (420 hp)
           With controllable metal propeller
           Maximum, except take-off
(Below 4000 ft.) 28.5 in. Hg., 2200 rpm (330 hp)
           (4000 ft. and up) 27.0 in. Hg., 2200 rpm (330 hp)
Take-off (one minute) 35.0 in. Hg., 2200 rpm (420 hp)
Eard speeds Level flight or climb - 206 mph True Ind.
      Placard speeds
                                          Glide or dive - 225 mph true Ind. Plaps extended - 100 mph True Ind.
                                          3915 lbs.
      Standard weight
                                           (+16.6) and (+29.0)
      C.G. limits
III - MODEL SC17R, 5 PCSB:
                                          Wright R-975E-3 or R-975E-2
      Engine
                                          Maximum, except take-off
      Placard limits
                                             34.5 in.Hg., 2200 rpm (420 hp)
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Propeller Placard speeds

Standard weight C.G. limits

Take-off (one minute)

35.0 in-Hg., 2200 rpm (420 hp)
Controllable metal only (Item 203)
Level flight or climt - 178 mph True Ind.
Glide or dive - 216 mph True Ind.
Flaps extended - 100 mph True Ind.
4105 lbs.
(+17.5) and (+25.9)